

GOES I-IM On-Orbit Storage





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GOES: Chronology of Coverage

NASA/NOAA GEOSYNCH CHRONOLOGY

ATS-1, 2, 3

SMS

GOES Spinners

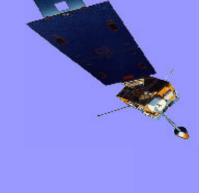
GOES D-H

GOES 3-Axis

GOES I-M

GOES N-Q

Advanced Studies



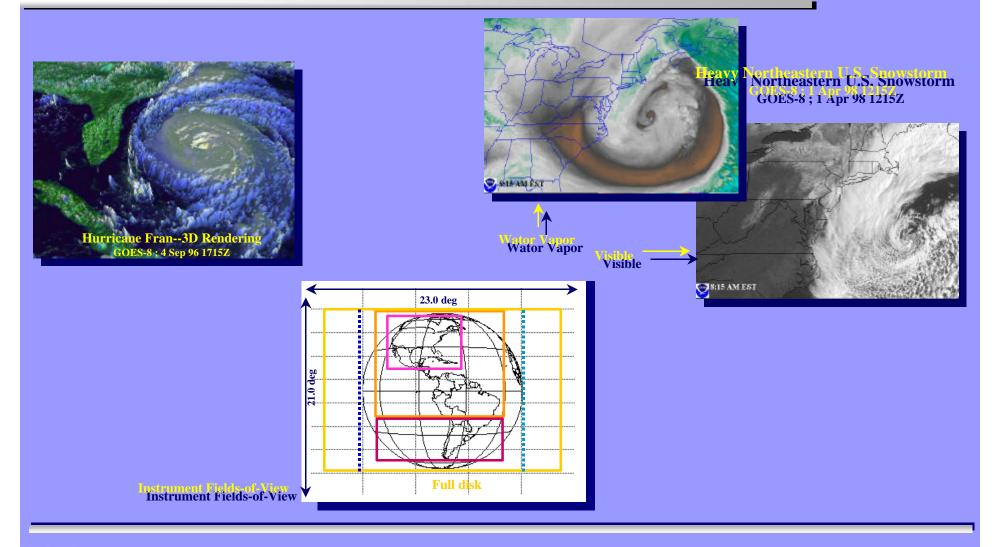








GOES Products for Forecasting

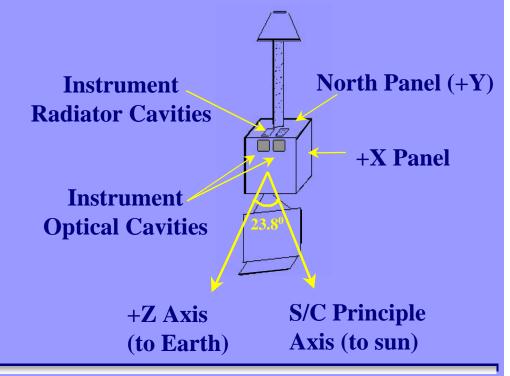






-Z Axis Precession (ZAP) Storage Spin-up & Dynamics

- Spacecraft powered down to eclipse configuration:
 - » Gyros (initial spin-up only)
 - » Attiude Control Electronics and Telemetry & Command equipment on
 - » Magnetic Torquers for additional spin control
 - » Heaters
- Solar Array slewed to 151⁰
 - » Two slews
 - » Sun Acquisition Mode commanded between slews
- Orient spacecraft principle axis to sun
- Coordinated thruster firings to initiate spin

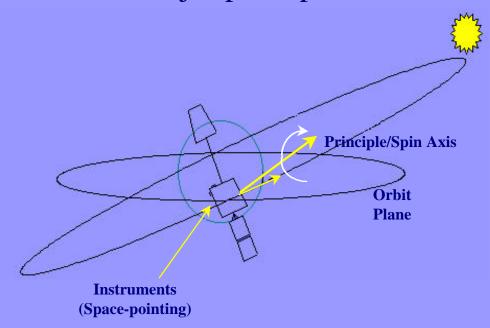






ZAP Mode Performance (1/2)

• 0.75% sec roll about major principle axis of inertia (23.8% off Z)



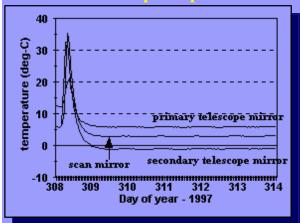
- Return to normal on-orbit mode extremely simple
 - » Sequence of sun acq/roll earth acq/station keeping utilized routinely
 - » Payload accomodations necessary before returning to operations



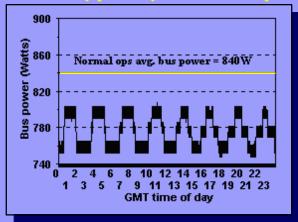


ZAP Mode Performance (2/2)

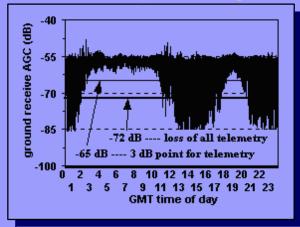
Instrument Telescope's rapid cool-down



Solar Array power cycles with ZAP spin



Ground AGC is cyclic in ZAP spin

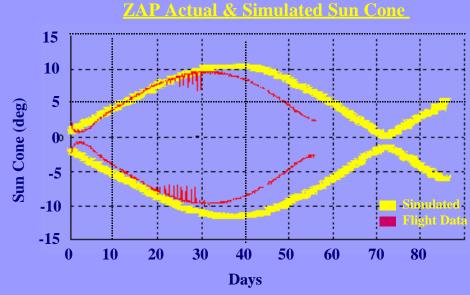






Spin Precession Requires Re-spins

- Principle Axis/Body Frame Axis angular seperation causes precession from initial coning angle relative to sun
- GOES-10 precessed at 0.40/day on initial activation
 - » At this rate, respin required every ~50 days
 - » Reductions possible with better Solar Array positioning relative to the sun
- Respin execution identical to initial spin but without Solar Array repositioning.
 - » First execution of respin successful
 - Precession rate reduced to
 0.35% (~ 60 day respin)







• Pseudo-telemetry "automates" monitoring

- T&C System (GIMTACS) algorithms developed and implemented for autonomous telemetry monitoring:
 - » Predict spin rate
 - » Predict T&C nulls, compare to actual ground system nulls; alarm if anomalous
 - » Predict sun coning angles from course sun sensor output and alarm if $>20^{\circ}$
 - » Predict eclipse entry & exit (automation of eclipse configurations)
 - » Detect anomalous solar array power and alarm if anomalous
 - » Detect loss of command link, and alarm
 - » Calculate spacecraft true local time
- 13 new pseudo-telemetry points added for these functions
- Staffing reduced from 3 full- to 1 part-time operators (per S/C)
 - Operational GOES Engineer monitors Storage S/C console
 - Full team required only for respins





- Pseudo-TLM points reduce real-time monitoring
- All TLM archived for analysis/trending
- No immediate action contingencies eliminate quick reaction responses
- Minimal ground resources required, leaving redundancy intact
 - One ground antenna (only for command link testing)
 - One GIMTACS stream







ZAP STORAGE PROVIDES FLEXIBILITY

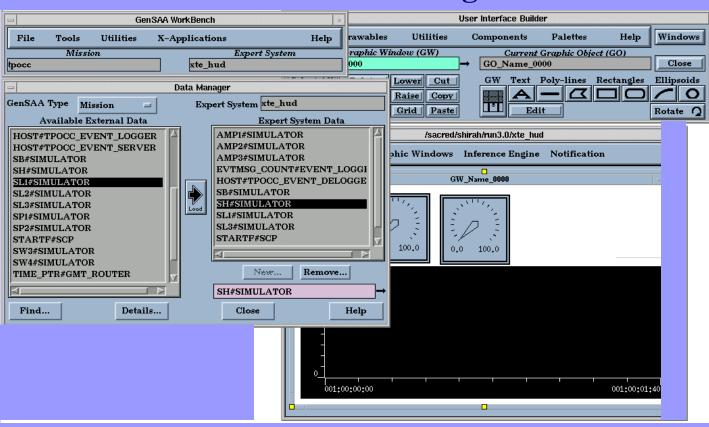
- On-orbit asset state-of-health and launch ques may result in a fourth GOES on orbit next year
- GOES-L (to be GOES-11) can be stored with minimal impact to ground resources
- Required operating mode of GOES N-Q (next generation)
- Incorporation of an Expert System for satellite state-of-health modelling will assist on-line operators in monitoring multiple stored spacecraft, with added safety





The Future of GOES On-Orbit Storage (2/3)

Generic Spacecraft Analyst's Assistant (GenSAA) for nearterm automation of on-orbit storage



GenSAA Data Manager & User Interface





The Future of GOES On-Orbit Storage (3/3)

